

Beyond Kalman Filtering: the TKF

by

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ABSTRACT

In automatic control theory, Kalman's important *Separation Principle* enables the designer to first design an optimal State-Feedback Control System and, secondly, an optimal (or *unbiased minimal-variance*) State Estimator or **Kalman Filter (KF)**, and then achieve by their combination the same as if one had used Stochastic Optimal Control Theory *ab initio*; in guidance systems this is called the *Guidance/Navigation Separation Principle* [1]. But in pure Navigation (or Tracking) Systems one does *not* always need an optimal estimate of the entire n -dimensional state-vector $x = [x_1, x_2, \dots, x_n]'$; for example, one may need only an unbiased minimal-variance estimate of the first component x_1 of x . In such cases one may use a **Trans-Kalman Filter (TKF)** which obtains *smaller* values of the variance P_{11} of the x_1 -estimate than is obtainable by *any* KF [which requires minimization of $(P_{11} + P_{22} + \dots + P_{nn}) = \text{trace}(P)$ and of intrinsic necessity weights *identically* the variance of the estimate of *each* component of x].

Introduction

For simplicity, we shall provide details only in the case of Linear Time-Invariant (**LTI**) processes, although as usual the results may be generalized readily to include all near-to-equilibrium or near-nominal-path *non-stationary linearizeable* processes in the obvious manner, particularly if the system's *robustness margin* ρ as defined in Appendix 2 is adequately large. For further simplicity (and with an eye to the usual real-time implementation by digital computing) we suppose that the problem is considered only in Discrete Time, wherein the time-interval $dT \equiv t_k - t_{k-1}$ between epochs $\{t_k\}$ of process-output observations, or sensor measurements, need not be constant, but is considered to be *known retroactively* at each new observation's epoch t_k , for $k = 1, 2, 3, \dots$. Then, letting $x^k = x(t_k)$, ($k = 1, 2, 3, \dots$), one considers the LTI stochastic process

$$x^{k+1} = \Phi \cdot x^k + \Gamma \cdot v^k, \quad (1)$$

$$y^k = H \cdot x^k + w^k, \quad (2)$$

where Φ is a constant $n \times n$ *state-transition* matrix and Γ is a constant $n \times m$ *exogenous input-coupling* matrix or *actuator* matrix (with $m \leq n$), and H is a constant $l \times n$ *output coupling* matrix or *sensor* matrix (with $l \leq n$), and where the m -vector v^k and the l -vector w^k are, respectively, zero-mean Gaussian stochastic white-noise processes of given *covariance* matrices Q & R , *i.e.*

$$E\{v^k\} \equiv 0, \quad E\{v^j \cdot (v^k)'\} = \delta_{jk} \cdot Q, \quad (3)$$

$$E\{w^k\} \equiv 0, \quad E\{w^j \cdot (w^k)'\} = \delta_{jk} \cdot R, \quad (4)$$

where ' denotes vector-matrix *transposition*; E denotes mathematical *expectation*; δ_{jk} denotes the Kronecker delta; and $Q = Q' \geq 0$ is a non-negative definite $m \times m$ *disturbance-covariance* matrix, while $R = R' > 0$ is a positive-definite $l \times l$ *noise-covariance* matrix. (If there is also present a known input it can be removed from consideration in (1) by the well-known Algebraic Separation Principle [cf. e.g. [18], pp. 62-63].)

We shall assume that the time for the Estimator Algorithm to process an output measurement y^k is negligible in comparison to dT , so that the measurement of y^{k+1} may be used virtually instantaneously to provide an online real-time *estimate* \hat{x}^{k+1} of x^{k+1} . Accordingly we postulate the *information-theoretic architecture* of the KF, namely the *linear estimator* (see Rugh [2], Exercise 29.3, p. 566)

$$\begin{aligned}\hat{x}^{k+1} &= \Phi \cdot \hat{x}^k + K \cdot \{ y^{k+1} - H \cdot \Phi \cdot \hat{x}^k \} \equiv \\ &\equiv (I_n - K \cdot H) \cdot \Phi \cdot \hat{x}^k + K \cdot H \cdot \Phi \cdot x^k + K \cdot H \cdot \Gamma \cdot v^k + K \cdot w^{k+1},\end{aligned}\quad (5)$$

where K is the celebrated $n \times m$ *Kalman gain matrix* and which *estimator*, when subtracted from (1) to provide the dynamics of the *estimation-error* $\tilde{x} \equiv: x - \hat{x}$, yields (after substituting new stochastic processes $\{\xi^k\}$ and $\{\eta^k\}$ by the *definitions* $\xi^k \equiv: (I_n - K \cdot H) \cdot \Gamma \cdot v^k$, $\eta^k \equiv: -K \cdot w^{k+1}$)

$$\tilde{x}^{k+1} = \Psi \cdot \tilde{x}^k + \xi^k + \eta^k, \quad E\{\xi^k \cdot (\eta^k)'\} \equiv 0, \quad (6)$$

where

$$\Psi \equiv: (I_n - K \cdot H) \cdot \Phi, \quad E\{\xi^k\} \equiv 0, \quad E\{\eta^k\} \equiv 0, \quad (7a)$$

$$E\{\xi^j \cdot (\xi^k)'\} = \delta_{jk} \cdot \tilde{Q}, \quad \tilde{Q} \equiv: (I_n - K \cdot H) \cdot \Gamma \cdot Q \cdot \Gamma' \cdot (I_n - K \cdot H)', \quad (7b)$$

$$E\{\eta^j \cdot (\eta^k)'\} = \delta_{jk} \cdot \tilde{R}, \quad \tilde{R} \equiv: K \cdot R \cdot K', \quad (7c)$$

and where I_n denotes the $n \times n$ *identity matrix*. Note that in the continuous-time KF one must allow for the possibility of a correlation between disturbance $v(t)$ and noise $w(t)$, but in discrete time this complication happily disappears! Now define P_k , the $n \times n$ *estimation-error covariance* matrix, by $P_k \equiv: E\{\tilde{x}^k \cdot (\tilde{x}^k)'\} \equiv (P_k)' \geq 0$, and as usual find that, for $k = 1, 2, 3, \dots$,

$$P_{k+1} = \Psi \cdot P_k \cdot \Psi' + \tilde{Q} + \tilde{R}. \quad (8)$$

In the LTI case it is convenient to take K to be a constant, so we postulate that P_k has a steady-state limit $P = P' \geq 0$, i.e. $P_k \rightarrow P$ as $k \rightarrow +\infty$. Thus, finally, we have

$$P = \Psi \cdot P \cdot \Psi' + \tilde{Q} + \tilde{R}. \quad (9)$$

For convenience, convert this discrete-time *Lyapunov Equation* (9) to its continuous-time equivalent (cf. Grewal & Andrews [3], pp. 133-152 [assuming their $H \equiv 0$ to reduce their Riccati equation to our Lyapunov equation], or J.N. Little's MATLAB program *dlyap* [4]):

$$\Psi_c \cdot P + P \cdot \Psi_c' + \tilde{Q}_c + \tilde{R}_c = 0, \quad (10a)$$

$$\Psi_c \equiv: (\Psi + I_n)^{-1} \cdot (\Psi - I_n), \quad (10b)$$

$$\tilde{Q}_c \equiv: (1/2)(I_n - \Psi_c) \cdot \tilde{Q} \cdot (I_n - \Psi_c'), \quad (10c)$$

$$\tilde{R}_c \equiv: (1/2)(I_n - \Psi_c) \cdot \tilde{R} \cdot (I_n - \Psi_c'). \quad (10d)$$

Let the linear operator vec be defined as taking an $n \times n$ matrix M and stacking its columns one above the other, in sequential order, to define an n^2 -dimensional vector $\mathbf{m} \equiv: vec(M)$, and define the unique inverse linear operator mat such that $M \equiv: mat(\mathbf{m}) \equiv mat(vec(M))$. Also let \otimes denote the Kronecker product (cf. Bellman [4], p. 239), so that if $I_n \otimes \Psi_c$ has no repeated eigenvalues then the unique solution of the Lyapunov equation (10a) is given by

$$P = vec(\mathbf{p}), \quad (11a)$$

where \mathbf{p} is found by solving the non-singular system of linear equations [equivalent to (10a)]

$$[(\Psi_c \otimes I_n) + (I_n \otimes \Psi_c')] \cdot \mathbf{p} = -vec(\tilde{Q}_c) - vec(\tilde{R}_c). \quad (11b)$$

The net effect of the preceding is to define uniquely scalar functions $\tilde{\phi}$, $\tilde{\psi}$ such that

$$\tilde{\phi} = \tilde{\phi}(K, Q; \Phi, \Gamma, H, dT), \quad \tilde{\psi} = \tilde{\psi}(K, R; \Phi, \Gamma, H, dT), \quad (12a)$$

$$P_{11} = \tilde{\phi}(K, Q) + \tilde{\psi}(K, R). \quad (12b)$$

For example, if $l = m = 1$, then

$$Q = Q_0 \cdot \sigma_d^2 > 0, \quad R = R_0 \cdot \sigma_s^2 > 0, \quad (13a)$$

where $Q_0 = Q_0(\Phi, \Gamma, H, dT)$, $R_0 = R_0(\Phi, \Gamma, H, dT)$, and where σ_d^2 is the continuous-time process-disturbance spectral density and σ_s^2 is the continuous-time sensor-noise spectral density, and we may therefore find scalar functions ϕ and ψ such that, by repetition of the preceding arguments,

$$\phi = \phi(K; \Phi, \Gamma, H, dT), \quad \psi = \psi(K; \Phi, \Gamma, H, dT), \quad (13b)$$

$$P_{11} = \phi(K) \cdot \sigma_d^2 + \psi(K) \cdot \sigma_s^2, \quad (13c)$$

where by inspection of (7)-(13) it may be verified that P_{11} is truly *linear* in the disturbance and sensor variances σ_d^2 & σ_s^2 as claimed.

Problem Statement

In general (at least, after a preliminary de-correlation procedure [3]), Q and R are diagonal in their elements $\{Q_{jj}\}$ and $\{R_{kk}\}$. Specifically, let $\{e^j\}$ & $\{u^k\}$ denote the unit column-vectors of I_m and I_l , and form the corresponding dyads $E^j = e^j \cdot (e^j)'$, ($j = 1, 2, \dots, m$) and $U^k = u^k \cdot (u^k)'$, ($k = 1, 2, \dots, l$), so that Q is a sum of m terms $E^j \cdot Q_{jj}$ and, similarly, R is a sum of l terms $U^k \cdot R_{kk}$ where each matrix E^j & U^k has only 0's and 1's as its elements. Then a repetition of the preceding arguments, *mutatis mutandis*, establishes via linearity the existence of m scalar functions $\{\phi_j(K) \mid (j = 1, 2, \dots, m)\}$ and l scalar functions $\{\psi_k(K) \mid (k = 1, 2, \dots, l)\}$ such that

$$P_{11} = \sum_{j=1}^m \phi_j(K) \cdot Q_{jj} + \sum_{k=1}^l \psi_k(K) \cdot R_{kk}. \quad (13d)$$

The TKF introduced here consists of *choosing* K to *jointly minimize* the coefficients $\{\phi_j(K)\}$ and $\{\psi_k(K)\}$ which display P_{11} as a linear function of the diagonal elements of Q & R . However, physically realistic values of the process-disturbance variances Q_{jj} and the sensor-noise variances R_{kk} are seldom if ever available. Accordingly it is common practice in the aerospace industry to “tune the KF” by empirical adjustment of Q & R until satisfactory performance is predicted via computer simulations (prior to actual estimator-system implementation, test, development, and deployment).

In the classical KF, not merely P_{11} but $(P_{11} + P_{22} + \dots + P_{nn}) = \text{trace}(P)$ is minimized by the choice of K , but (although this yields a stable system) it does NOT make P_{11} alone as small as it can be made when some *tolerable* degradations of the remaining P_{kk} are allowed, and there is no way short of an obvious elaboration of the theory developed herein [using (11b) and the individual parallels to (13d) for each of the remaining P_{kk}] to minimize a *weighted* sum of the P_{kk} with arbitrary weights.

Also, unfortunately, it has been found in practice that the classical KF is often quite *fragile* in the sense that $\sigma^2 = \text{trace}(P)$ as a function of $(\{Q_{jj}\}, \{R_{kk}\})$ can have a very sharply defined minimum with severely steep walls so that when even a slightly off-nominal pair (Q, R) is encountered the resultant change in K may cause the filter’s performance to degrade unacceptably from that expected. This has led to much study of possible methods to render KFs more *robust* (e.g. [6]-[9]), including a recent generalization to discrete time of the 1956 Bass “rhubustness” criterion ρ [10] presented succinctly in Appendix 2 below. But in the present context, all such efforts have hitherto suffered from the problem presented in the preceding paragraph.

Therefore we introduce here the **TKF** as an alternative approach to attempt to render the system *maximally robust* to unknown/unpredictable deviations in Q & R (which also to a considerable extent can compensate for un-modeled cross-couplings and/or discrepant deviations from the assumed design-nominal parameter-values in Φ , Γ , and H , as quantified in Appendix 2).

New Resolution

The Trans-Kalman Filter (**TKF**) is one in which the $(l \cdot n)$ distinct parameter-values defining the Kalman gain matrix K are so chosen as to make the $(l + m)$ nonlinear functions $(\{\phi_j(K)\}, \{\psi_k(K)\})$ *jointly* “as small as possible” in an appropriate sense (that in each particular application may be defined on an *ad hoc* basis).

To illustrate this point in the context of the simplest possible TKF, wherein $n = 2$, and as already considered above, $l = m = 1$, consider the tracking problem ([11] – [17]) and in particular the widely used steady-state/limiting KF called an Alpha-Beta Tracker. Today’s computers are so powerful and cost-effective that it is practical to use real-time nonlinear trigonometric transformations to resolve measured range, azimuth and inclination into rectilinear Euclidean coordinates, in which the dynamics of a point-particle moving in a straight line at constant velocity become uncoupled, so that two (or, in the case of a change of altitude, three) *one-dimensional* tracking problems may be considered independently. Thus, consider rectilinear constant-speed motion defined by a state-vector $x = [r, v]^T$, where r denotes range and $v = dr/dt$

denotes velocity, and let r_m denote the measured range. Then the α - β Tracker Algorithm may be stated as

$$r_{\text{new}} = (1 - \alpha) \cdot [r_{\text{old}} + v_{\text{old}} \cdot dT] + \alpha \cdot r_m, \quad (0 < \alpha < 2), \quad (14a)$$

$$v_{\text{new}} = (1 - \beta) \cdot v_{\text{old}} + \beta \cdot [(r_m - r_{\text{old}})/dT], \quad (0 < \beta < 2 \cdot (2 - \alpha)). \quad (14b)$$

Note that even if α & β are taken to be constants, the algorithm (14) automatically *adapts* to whatever the most recent update time-interval $dT \equiv: t_{\text{new}} - t_{\text{old}}$ may have been. Moreover, for fixed dT , (14) may be regarded as a steady-state or limiting KF by setting

$$\Phi = \begin{bmatrix} 1 & dT \\ 0 & 1 \end{bmatrix}, \quad \Gamma \cdot Q \cdot \Gamma' = \begin{bmatrix} dT^3/3 & dT^2/2 \\ dT^2/2 & dT \end{bmatrix} \cdot \sigma_a^2, \quad (15)$$

$$H = [1, 0], \quad K = [\alpha, \beta/dT]', \quad R = \sigma_s^2/dT, \quad (16)$$

where σ_a^2 is the continuous-time variance of white-noise-like acceleration disturbance and σ_s^2 is the continuous-time variance of the white-noise-like sensor noise (as explained and derived by Kamen & Su [17], pp. 319-22). In this $n = 2$ case of (15)-(16), straightforward application of the preceding theory (1)-(11) yields, after eliminating the redundancy in the $n^2 = 4$ equations in (11), just $(1/2) \cdot n \cdot (n+1) = 3$ independent linear equations in 3 unknowns, to solve (9) for (P_{11}, P_{12}, P_{22}) . After a truly enormous amount of algebra, best done by a symbolic-algebra engine, the final result simplifies greatly down to

$$P_{11} = \phi(\alpha, \beta) \cdot \sigma_a^2 \cdot (dT^3/3) + \psi(\alpha, \beta) \cdot \sigma_s^2/dT, \quad (17a)$$

$$\phi \equiv: [(1 - \alpha)^2/\alpha] \cdot \{[3 \cdot (2 - \alpha) - \beta]/[\beta \cdot (2 \cdot (2 - \alpha) - \beta)]\}, \quad (17b)$$

$$\psi \equiv: [2 \cdot \alpha^2 + (2 - 3 \cdot \alpha) \cdot \beta]/[\alpha \cdot (2 \cdot (2 - \alpha) - \beta)], \quad (17c)$$

Here is ψ known in the literature [cf. e.g. [11], p. 27, eqn. 1.2-20] but ϕ seems to be a novel consequence of the more complete theory (13d) introduced herein. Taking σ_s^2 to be a constant, minimizing P_{11} is the same as minimizing $\tilde{P}_{11} \equiv: P_{11}/\sigma_s^2$ so, defining the disturbance-to-noise ratio ρ by

$$\rho \equiv: \sigma_a^2/\sigma_s^2, \quad (18)$$

we wish to minimize

$$\tilde{P}_{11} = \phi(\alpha, \beta) \cdot \rho \cdot (dT^3/3) + \psi(\alpha, \beta)/dT. \quad (19)$$

For each dT and for each “physically plausible” value of the ratio ρ , we can use standard numerical methods to find the absolute minimum of \tilde{P}_{11} on the sub-triangular (α, β) -domain of assured positivity of \tilde{P}_{11} , namely $\{0 < \alpha < 1, 0 < \beta < 2 \cdot (2 - \alpha)\}$, in which case it turns out that plotting α and β versus dT reveals that they are essentially constant for $dT \geq 1$. Accordingly one may try the “robustification” approach ([6]-[10]) by allowing ρ and dT to be

arbitrary unknown parameters and choosing constant values of (α, β) to “jointly” minimize (ϕ, ψ) in an appropriate sense.

By inspection of (17b) it makes no sense to vary α to minimize ϕ because this immediately yields $\alpha \equiv 1$, which is tantamount to not using an α - β Tracker! [See Appendix 1 for more detailed proof.] Accordingly we regard α as fixed and focus on minimizing

$$\hat{\phi} \equiv: [3 \cdot (2 - \alpha) - \beta] / [\beta \cdot (2 \cdot (2 - \alpha) - \beta)], \quad (20)$$

Differentiating with respect to β and simplifying, one finds

$$\begin{aligned} \beta^2 \cdot [2 \cdot (2 - \alpha) - \beta]^2 \cdot (\partial \hat{\phi} / \partial \beta) &= -\{\beta^2 - 6 \cdot (2 - \alpha) \cdot \beta + 6 \cdot (2 - \alpha)^2\} \equiv \\ &\equiv -(2 - \alpha)^2 \cdot (\kappa^2 - 6 \cdot \kappa + 6) \equiv \\ &\equiv -(2 - \alpha)^2 \cdot (\kappa - 3 + 3^{1/2}) \cdot (\kappa - 3 - 3^{1/2}) = 0, \end{aligned} \quad (21)$$

wherein we were required [by inspection of the preceding quadratic form in β and the hypothesis $(\partial \hat{\phi} / \partial \beta) = 0$], to have assumed that β has the form

$$\beta = \hat{\beta}(\alpha) \equiv: \kappa \cdot (2 - \alpha), \quad \kappa = 3 - 3^{1/2} = 1.267949192, \quad (22)$$

because the other possible choice of κ make β too large. Accordingly the choice (22) minimizes the consequences of large ρ or large dT regardless of what α -value may have been selected! Substituting (22) into

$$\hat{\psi} = \hat{\psi}(\alpha) \equiv: \psi(\alpha, \hat{\beta}(\alpha)) \quad (23)$$

yields, after differentiation and simplification,

$$(1/2) \cdot (2 - \kappa) \cdot (2 - \alpha)^2 \cdot \alpha^2 \cdot (\partial \hat{\psi} / \partial \alpha) = (2 - \kappa) \cdot \alpha^2 + 4 \cdot \kappa \cdot \alpha - 4 \cdot \kappa = 0, \quad (24)$$

whence minimization of $\hat{\psi}$ is achieved by the choice [at first, for arbitrary κ]

$$\alpha = 2 \cdot (-\kappa + [2 \cdot \kappa]^{1/2}) / (2 - \kappa) = 0.886553879, \quad (25)$$

which shows, for *any* κ , that $\hat{\psi}(\alpha) \equiv \alpha$ [!], and then from (22), also implies that

$$\beta = 1.411793110. \quad (26)$$

Conclusion

The *maximally robust* (or, in complete generality, “least worst”) α - β Tracker is given by

$$\alpha_{\text{OPT}} = 0.886554, \quad \beta_{\text{OPT}} = 1.411793, \quad (27)$$

$$P_{11} = (0.0243289) \cdot \sigma_a^2 \cdot (dT^3/3) + \alpha_{\text{OPT}} \cdot \sigma_s^2 / dT, \quad (28)$$

as shown in an Numerical Appendix In this case the error-state *closed-loop* state-transition matrix becomes

$$\Psi = \begin{bmatrix} 1-\alpha & (1-\alpha) \cdot dT \\ -\beta/dT & (1-\beta) \end{bmatrix}, \quad (29)$$

whose characteristic polynomial turns out, remarkably, to be independent of dT [thus further justifying the choice of constant (α, β)], and is

$$z^2 - (2 - \alpha - \beta) \cdot z + (1 - \alpha) \equiv (z - z_1) \cdot (z - z_2) = 0, \quad (30)$$

which has complex roots $z_{1,2} \equiv: -0.149 \pm 0.302 \cdot i$ that satisfy [cf. Numerical Appendix & λ in Appendix 2]

$$|z_1| = |z_2| = 0.336818 \quad (31)$$

and therefore define poles *inside* the unit circle of the complex z -plane, ensuring tracker stability for arbitrary but known dT and *arbitrary/unknown* (σ_a^2, σ_s^2) . Indeed, by the known results [2] cited in Appendix 2, the state-vector estimate's error-norm $\|\tilde{x}^k\|$ decreases by a factor of at least $(1/3)$ [e.g. < 0.04 for $k \geq 3$] after each new update at the same time-increment dT !

Appendix 1

On second thought, perhaps we should have considered the possibility of taking $\alpha = 1$. But in that case, since $\phi \equiv 0$ and $\psi(1, \beta) \equiv [(2 - \beta)/(2 - \beta)] \equiv 1$, we find that $P_{11} = \sigma_s^2/dT$, whence the value of σ_a^2 is irrelevant and β may be chosen arbitrarily, subject only to $0 < \beta < 2$. But in the event of negligible σ_a^2 and/or small dT ($<< 1$), this result is clearly inferior to (28), so that in hindsight the decision to ignore the factor $[(1 - \alpha^2)/\alpha]$ in ϕ was correct after all.

Appendix 2

A STABILITY-RHOBUSTNESS MARGIN IN DISCRETE TIME

by

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Excerpt from a letter to Prof. Wilson J. Rugh of the Dept. of Electrical & Computer Engineering at the Johns Hopkins University dated August 13, 2002 in which I told him about my having generalized from analog time to discrete-time (in connection with an evening course on DSP for the local branch of Florida Inst. of Technology that I was then teaching) my 'Rhubustness' Criterion that I had first published for analog-time systems in 1956 [10] and then improved in an IEEE Conference paper [8] in 1991):

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I have just derived what I believe is an elegant generalization from continuous-time systems to discrete-time systems of my theory of "rhubustification" of control and observer systems, as follows:

In your book you prove that if the system

$$x^{k+1} = A \cdot x^k + B \cdot u^k, \quad u = -K \cdot x,$$

is stabilized by state-feedback, to become [closed-loop]

$$x^{k+1} = A_{cl} \cdot x^k, \quad x^0 = x^o, \quad A_{cl} \equiv: A - B \cdot K,$$

then there must be positive numbers $\gamma \geq 1$ and $\lambda < 1$ [$\lambda \equiv: \max\{|\text{eig}(A_{cl})|\}$] such that

$$\|x^k\| \leq \gamma \cdot \|x^o\| \cdot \lambda^k, \quad (k = 1, 2, 3, \dots).$$

I now define a *stability “robustness” margin* ρ , ($0 < \rho < 1$), by $\rho \equiv (1 - \lambda)/\gamma$, and prove that if in **NONSTATIONARY/NONLINEAR & EXTERNALLY-FORCED** actuality the system is better-modeled by the 'perturbed' system

$$x^{k+1} = A_c \cdot x^k + f(k, x^k) + g(k), \quad (k = 1, 2, 3, \dots),$$

where there exist (κ, δ) such that, for all x and k , $\|f(k, x)\| \leq \kappa \cdot \|x\|$, $\|g(k)\| \leq \delta$, and where $\kappa < \rho$, then

$$\|x^k\| \leq \gamma \cdot \|x^0\| \cdot \Lambda^k + \delta/\rho, \quad \Lambda \equiv \{\lambda + [\kappa/\rho](1 - \lambda)\} < 1, \quad (k = 1, 2, 3, \dots).$$

Hence the *larger* is ρ , the **LESS SENSITIVE** is the actual system to *UNMODELLED cross-couplings* & other *NONSTATIONARY* and/or *NONLINEAR* effects, as well as *EXTERNAL FORCING!*

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